

6. Dresdner Probabilistic-Workshop 2013, 10<sup>th</sup>/11<sup>th</sup> October 2013

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# Enhanced Quadrature Approach applied to a Robust Compressor and Turbine Blade Design

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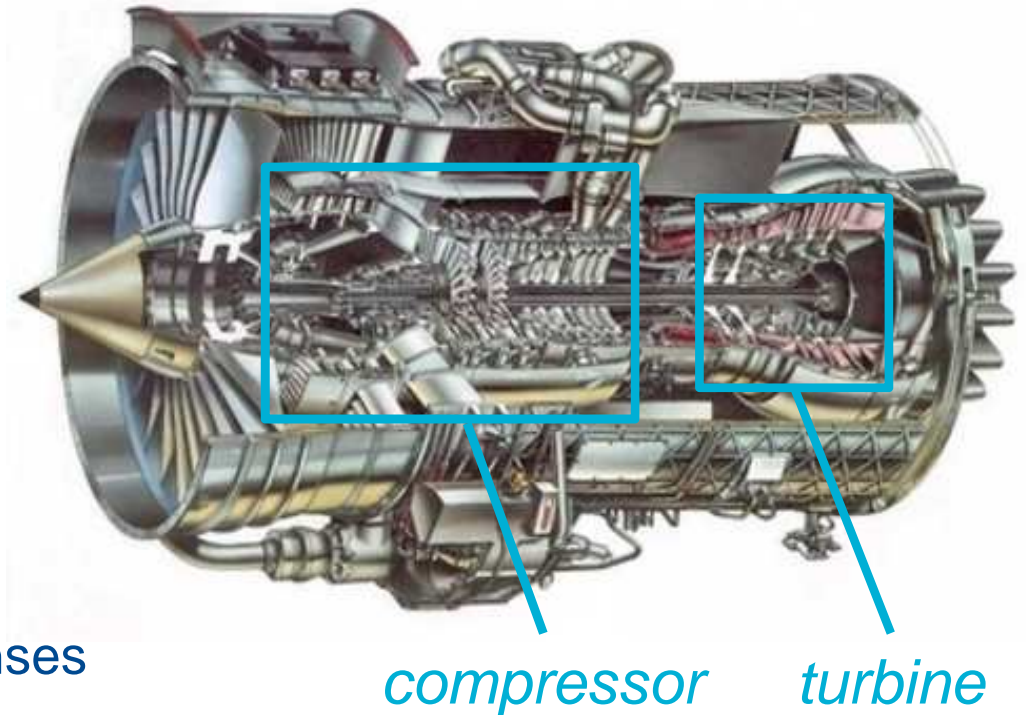
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- 6) Summary of the URQ method



# Motivation

- Problem:
  - Deterministic optimization does not feature manufacturing noise and degradation
- Solution:
  - probabilistic simulation describing variations of input parameters and corresponding system responses
- Opportunity :
  - Monte Carlo Simulation (DS, LHC, oLHC ) → very expensive
  - alternativ approach: **Univariate Reduced Quadrature (URQ)** [1]



[1] M. Padulo, M.S. Campobasso, and M.D. Guenov. Novel Uncertainty Propagation Method for Robust Aerodynamic Design. *AIAA Journal*, 49(3):530-543, 2011.

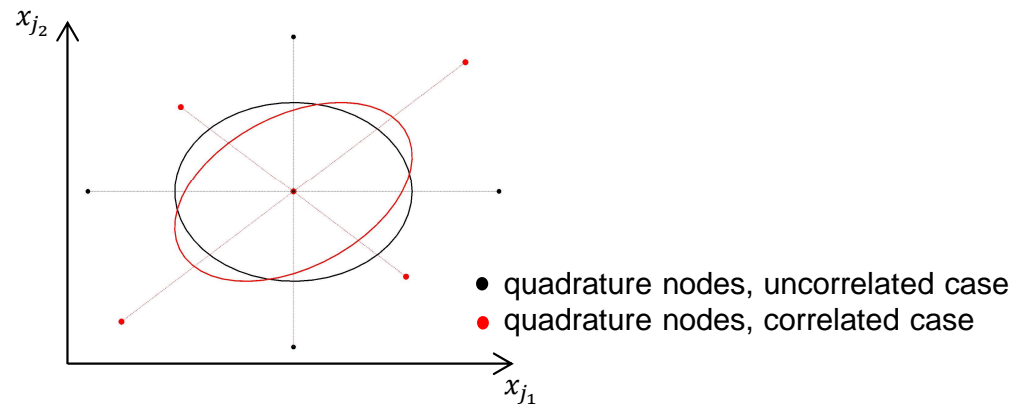
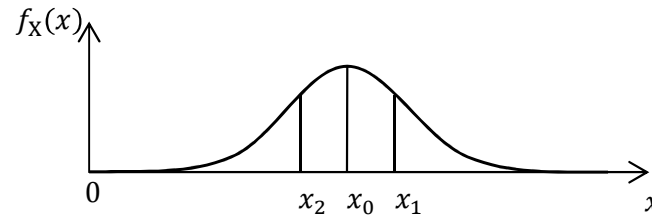


# Introduction to the URQ approach

- Subject: Finding an alternative approach, in comparison to expensive Monte Carlo Simulations to save calculation time for estimating  $\mu_g = E[g(x)]$  and  $\sigma_g^2 = \text{Var}[g(x)]$
- Basic Idea: Choose a deterministic approach that's based on a Gaussian-Quadrature with 3 nodes per dimension  $n_d$

$$\begin{aligned}x_0 &= \mu_X \\x_1 &= \mu_X + h_1 \sigma_X \\x_2 &= \mu_X + h_2 \sigma_X\end{aligned}$$

$$\begin{aligned}x_0 &= \mu_X \\x_{1_i} &= \mu_X + h_{1_i} \mathbf{C} \mathbf{e}_i \\x_{2_i} &= \mu_X + h_{2_i} \mathbf{C} \mathbf{e}_i\end{aligned}$$



Build the density function  $f_X(x)$  of the uncertain input parameter  $x$  with 3 deterministic chosen nodes  $x_0$ ,  $x_1$  and  $x_2$  for each dimension

# Introduction to the URQ approach

## ➤ Idea of URQ:

- build  $2n_d + 1$  nodes  $\mathbf{x}$  with corresponding weights  $\mathbf{w}$

$$\begin{aligned} \mathbf{x}_0 &= \boldsymbol{\mu}_X & \sigma_X & \dots \text{ standard deviation of } X \\ \mathbf{x}_{1_i} &= \boldsymbol{\mu}_X + h_{1_i} \sigma_{X_i} \mathbf{e}_i & h_{1,2} & \dots \text{ scalingparameter} \\ \mathbf{x}_{2_i} &= \boldsymbol{\mu}_X + h_{2_i} \sigma_{X_i} \mathbf{e}_i \end{aligned}$$

- propagate each node through the response function  $g(\mathbf{x})$  of interest
- build mean of system response by

$$\mu_g^{(URQ)} = w_0^{(\mu)} g(\mathbf{x}_0) + \sum_{i=1}^{n_d} \left\{ w_{1_i}^{(\mu)} g(\mathbf{x}_{1_i}) + w_{2_i}^{(\mu)} g(\mathbf{x}_{2_i}) \right\}$$



Calculation cost of  $2n_d + 1$  calculations



### Question

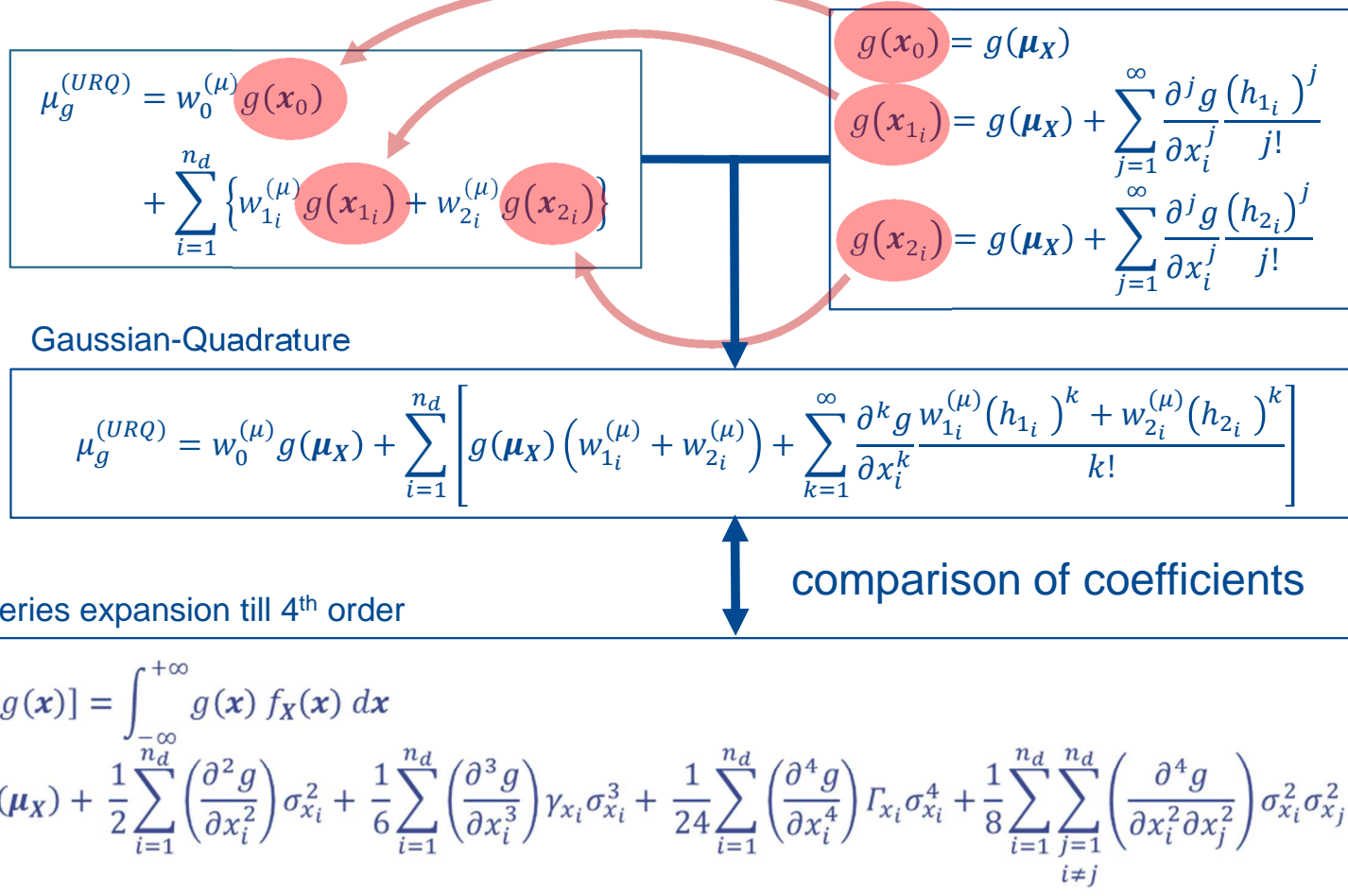
How to find optimal weights  $\mathbf{w}$  and scaling parameters  $\mathbf{h}$



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# Introduction to the URQ approach

- **Solution:** comparison of coefficients between  
Gaussian-Quadrature and Taylor-series expansion



# Introduction to the URQ approach

- System of eight equations, its solution leads to:
- Optimal values of scaling parameter  $h$

$$h_{1i} = \frac{\gamma_{X_i}}{2} + \sqrt{\Gamma_{X_i} - \frac{3\gamma_{X_i}^2}{4}}$$

$$h_{2i} = \frac{\gamma_{X_i}}{2} - \sqrt{\Gamma_{X_i} - \frac{3\gamma_{X_i}^2}{4}}$$

$\gamma_X$  ... skewness of  $X$   
 $\Gamma_X$  ... kurtosis of  $X$

- Optimal values of weights  $w$

Mean

$$w_0^{(\mu)} = 1 + \sum_{i=1}^{n_d} \frac{1}{h_{1i}h_{2i}}$$

$$w_{1i}^{(\mu)} = \frac{1}{h_{1i}(h_{1i} - h_{2i})}$$

$$w_{2i}^{(\mu)} = -\frac{1}{h_{2i}(h_{1i} - h_{2i})}$$

Variance

$$w_{0i}^{(\sigma)} = \frac{2}{h_{1i}h_{2i}(h_{1i} - h_{2i})^2}$$

$$w_{1i}^{(\sigma)} = \frac{(h_{1i})^2 - h_{1i}h_{2i} - 1}{(h_{1i})^2(h_{1i} - h_{2i})^2}$$

$$w_{2i}^{(\sigma)} = \frac{(h_{2i})^2 - h_{1i}h_{2i} - 1}{(h_{2i})^2(h_{1i} - h_{2i})^2}$$



# Introduction to the URQ approach

## ➤ Correlation of input parameters

- spectral decomposition:

$$\Sigma_{XX} = VDV^{-1}$$

- $\Sigma_{XX}$  ... input covariance matrix
- $V$  ... right eigenvector of  $\Sigma_{XX}$
- $D$  ... diagonal matrix of eigenvalues of  $\Sigma_{XX}$

$$C = V\sqrt{D}$$

- URQ nodes with correlation:

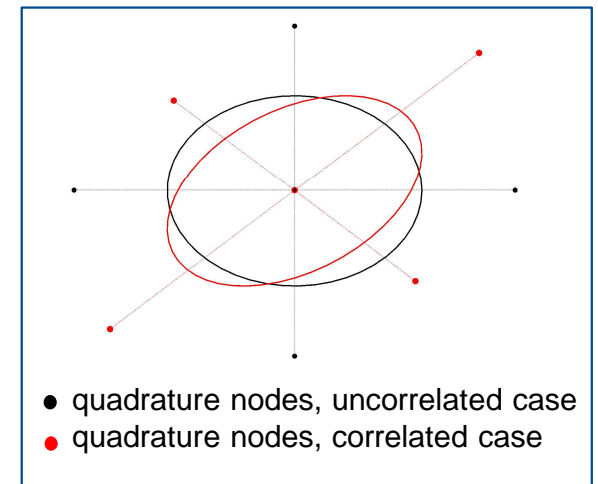
uncorrelated

$$\begin{aligned} x_0 &= \mu_X \\ x_{1i} &= \mu_X + h_{1i} \sigma_{X_i} e_i \\ x_{2i} &= \mu_X + h_{2i} \sigma_{X_i} e_i \end{aligned}$$



correlated

$$\begin{aligned} x_0 &= \mu_X \\ x_{1i} &= \mu_X + h_{1i} C e_i \\ x_{2i} &= \mu_X + h_{2i} C e_i \end{aligned}$$



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# Introduction to the URQ approach

This approach is called the

- Univariate** (no dependencies between parameters - only one dimension is altered to obtain each node → Spectral decomposition)
- Reduced** (as few nodes as possible, Taylor is truncated)
- Quadrature** (approximation of an integral via a sum)

**Method**



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# Performance of the URQ

- Compare URQ and oLHC by using CFD example
- Performance:

- DoE with 19 different compressor blade profiles
- each profile is defined by 14 deterministic design parameters

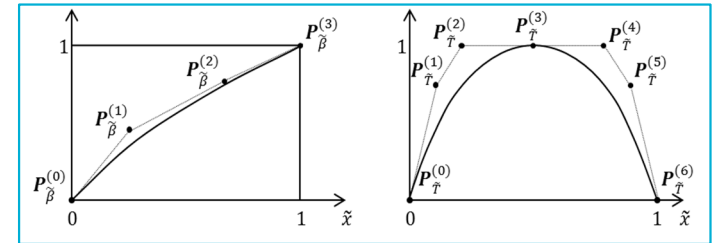
$$\mathbf{p} = [p_1 \dots p_{14}]^T \in \mathbb{R}^{14}$$

- uncertainty defined with 8 parameter  $\rightarrow n_d = 8$

$$\mathbf{d} = [\gamma \quad c \quad T_{max} \quad x_{T_{max}} \quad F_{max} \quad x_{F_{max}} \quad T_{LE} \quad T_{TE}] \in \mathbb{R}^8$$

where  $\mathbf{d} = \mathbf{d}(\mathbf{p})$

- determine following system responses:
  - mean of pressure loss
  - variance of pressure loss
  - mean of exit flow angle
  - variance of exit flow angle

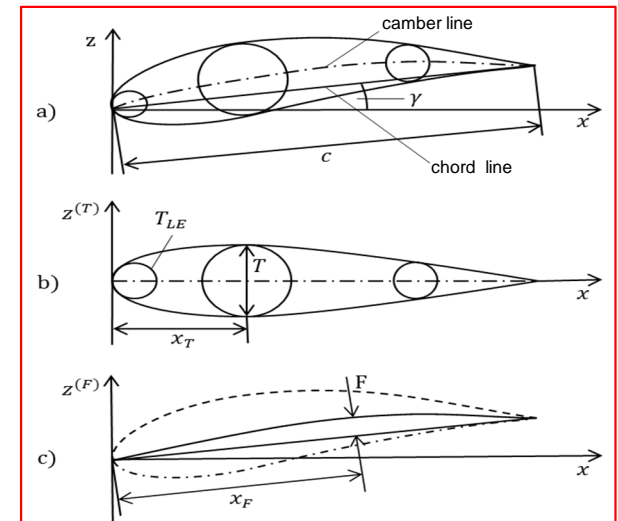


$$t = \frac{T_{max}}{c}$$

$$p_{12} = \frac{\frac{t}{c} - t_{min}}{t_{max} - t_{min}}$$

$$p_{13} = \frac{\beta_{LE} - \beta_{LEmin}}{\beta_{LEmax} - \beta_{LEmin}}$$

$$p_{14} = \frac{\beta_{TE} - \beta_{TEmin}}{\beta_{TEmax} - \beta_{TEmin}}$$



Question: „How much additional cost is needed to generate equivalent estimates?“

# Performance of the URQ

## ➤ Accomplishment:

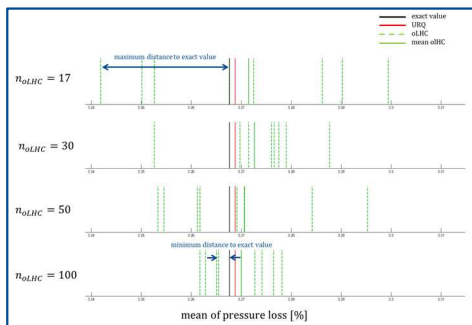
- reference (exact value) is oLHC with  $N = 10.000$
- number of nodes for URQ not varying:

$$n_{URQ} = 2n_d + 1 = 17$$

- number of nodes for oLHC varying:

$$n_{oLHC} = 17 \text{ (equivalent cost), } 30, 50 \text{ and } 100$$

- each oLHC- calculation was repeated 10x to show response variation :



→ 10 solution, out of that

- 1 run with minimal distance to optimal solution
- 1 run with maximal distance to optimal solution

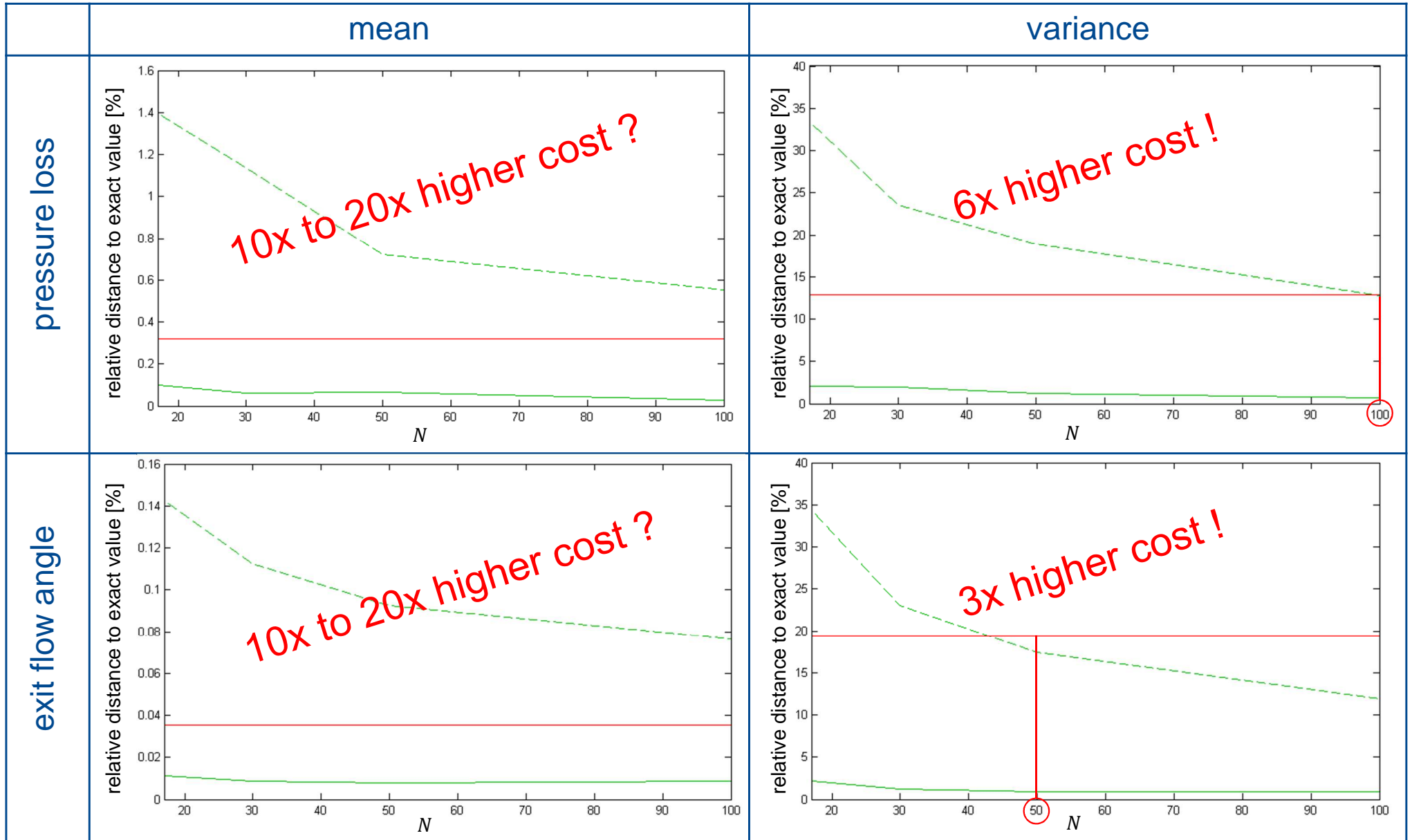
- build average out of all solutions of  $n_{DoE} = 19$  profiles
  - 1 average out of 19 profiles with minimal distance
  - 1 average out of 19 profiles with maximal distance



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# Performance of the URQ

- URQ
- oLHC (runs with min distance to optimal solution)<sup>12</sup>
- - - oLHC (runs with max distance to optimal solution)



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# Performance of the URQ

- 10- to 20-times higher cost for oLHC- simulation to get equivalent approximations for the mean of system response
- 3- to 6-times higher cost for oLHC- simulation to get equivalent approximations for the variance of system response

 Why is there a difference between the approximation- quality of mean and variance 

- **Reason:** Comparison of coefficients between Gaussian- Quadrature and Taylor-series expansion doesn't feature cross-derivative terms of the Taylor-series.



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# Performance of the URQ

## ➤ Approximations of system responses with Taylor-series

- Mean  $\mu_g$

$$\begin{aligned} \mu_g &= E[g(\mathbf{x})] = \int_{-\infty}^{+\infty} g(\mathbf{x}) f_{\mathbf{x}}(\mathbf{x}) d\mathbf{x} \\ &= \overbrace{g(\boldsymbol{\mu}_{\mathbf{x}})}^{M_1} + \overbrace{\frac{1}{2} \sum_{i=1}^{n_d} \left( \frac{\partial^2 g}{\partial x_i^2} \right) \sigma_{x_i}^2}^{M_2} + \overbrace{\frac{1}{6} \sum_{i=1}^{n_d} \left( \frac{\partial^3 g}{\partial x_i^3} \right) \gamma_{x_i} \sigma_{x_i}^3}^{M_3} + \overbrace{\frac{1}{24} \sum_{i=1}^{n_d} \left( \frac{\partial^4 g}{\partial x_i^4} \right) \Gamma_{x_i} \sigma_{x_i}^4}^{M_4} + \overbrace{\frac{1}{8} \sum_{i=1}^{n_d} \sum_{\substack{j=1 \\ i \neq j}}^{n_d} \left( \frac{\partial^4 g}{\partial x_i^2 \partial x_j^2} \right) \sigma_{x_i}^2 \sigma_{x_j}^2}^{M_5} + o(\sigma_{\mathbf{x}}^5) \end{aligned}$$

- Variance  $\sigma_g^2$

$$\begin{aligned} \sigma_g^2 &= E[(g(\mathbf{x}) - \mu_g)^2] = \int_{-\infty}^{+\infty} (g(\mathbf{x}) - \mu_g)^2 f_{\mathbf{x}}(\mathbf{x}) d\mathbf{x} \\ &= \overbrace{\sum_{i=1}^{n_d} \left( \frac{\partial g}{\partial x_i} \right) \sigma_{x_i}^2}^{S_1} + \overbrace{\sum_{i=1}^{n_d} \left( \frac{\partial^2 g}{\partial x_i^2} \right) \left( \frac{\partial g}{\partial x_i} \right) \gamma_{x_i} \sigma_{x_i}^3}^{S_2} + \overbrace{\sum_{i=1}^{n_d} \sum_{\substack{j=1 \\ i \neq j}}^{n_d} \left( \frac{\partial^3 g}{\partial x_i^2 \partial x_j} \right) \left( \frac{\partial g}{\partial x_j} \right) \sigma_{x_i}^2 \sigma_{x_j}^2}^{S_3} \\ &+ \overbrace{\frac{1}{2} \sum_{i=1}^{n_d} \sum_{\substack{j=1 \\ i \neq j}}^{n_d} \left( \frac{\partial^2 g}{\partial x_i \partial x_j} \right)^2 \sigma_{x_i}^2 \sigma_{x_j}^2}^{S_4} + \overbrace{\frac{1}{3} \sum_{i=1}^{n_d} \left( \frac{\partial^3 g}{\partial x_i^3} \right) \left( \frac{\partial g}{\partial x_i} \right) \Gamma_{x_i} \sigma_{x_i}^4}^{S_5} + \overbrace{\frac{1}{4} \sum_{i=1}^{n_d} \left( \frac{\partial^2 g}{\partial x_i^2} \right)^2 (\Gamma_{x_i} - 1) \sigma_{x_i}^4}^{S_6} + o(\sigma_{\mathbf{x}}^5) \end{aligned}$$



# Validation of the error

- Exact response mean and variance are calculated analytically for a test function together with a uniform probability distribution
- The input standard deviation is repeatedly halved
- Result: error ratios converge to 16  
→ Error of order  $O(\sigma_x^4)$

$\varepsilon_{\mu_f}(\sigma_x)$ ratio	$\frac{\varepsilon_{\mu_f}(0.32)}{\varepsilon_{\mu_f}(0.16)}$	$\frac{\varepsilon_{\mu_f}(0.16)}{\varepsilon_{\mu_f}(0.08)}$	$\frac{\varepsilon_{\mu_f}(0.08)}{\varepsilon_{\mu_f}(0.04)}$	$\frac{\varepsilon_{\mu_f}(0.04)}{\varepsilon_{\mu_f}(0.02)}$
Value	15.63	15.91	15.98	15.99
$\varepsilon_{\sigma_f^2}(\sigma_x)$ ratio	$\frac{\varepsilon_{\sigma_f^2}(0.32)}{\varepsilon_{\sigma_f^2}(0.16)}$	$\frac{\varepsilon_{\sigma_f^2}(0.16)}{\varepsilon_{\sigma_f^2}(0.08)}$	$\frac{\varepsilon_{\sigma_f^2}(0.08)}{\varepsilon_{\sigma_f^2}(0.04)}$	$\frac{\varepsilon_{\sigma_f^2}(0.04)}{\varepsilon_{\sigma_f^2}(0.02)}$
Value	15.34	15.83	15.96	15.99



# Aerodynamic blade-to-blade robust turbine blade optimization

## ➤ Problem definition:

$$\min_{d \in F} P(\omega > \omega^{nom})$$

subject to

$$P(|\beta_2 - \beta_2^{tar}| > \varepsilon) < p^{lim}$$

$$P(d > d^{lim}) < p^{lim}$$

$$P(H^{TE,SS} > H^{lim}) < p^{lim}$$

$$P(M_{max} > M^{lim}) < p^{lim}$$

$$\omega = \frac{p_{0,E}^{isen} - \bar{p}_{0,E}}{p_{0,I} - p_I}$$

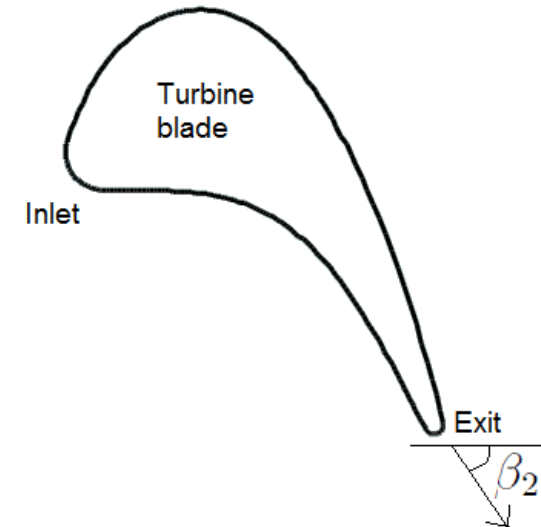
where

$p_{0,E}^{isen}$  ... isentropic total pressure at exit plane

$\bar{p}_{0,E}$  ... mass-averaged exit stagnation pressure

$p_{0,I}$  ... total pressure at inlet plane

$p_I$  ... static pressure at inlet plane

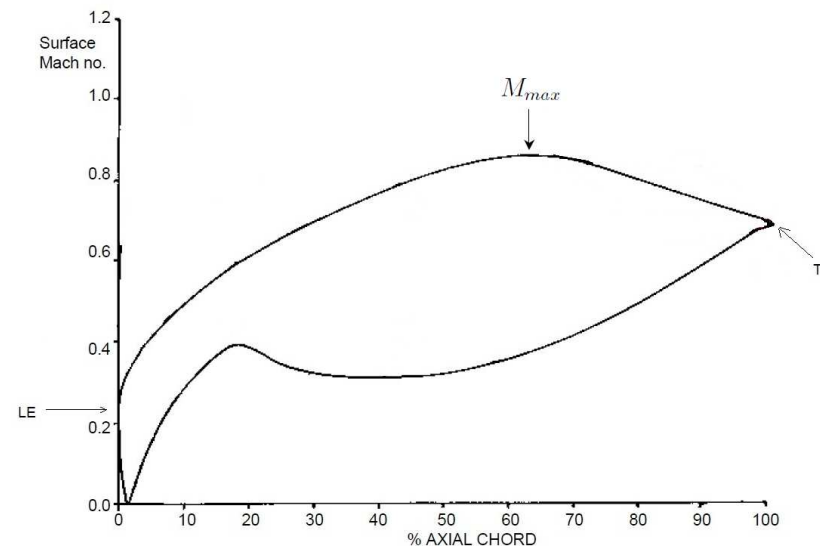


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# Aerodynamic blade-to-blade robust turbine blade optimization

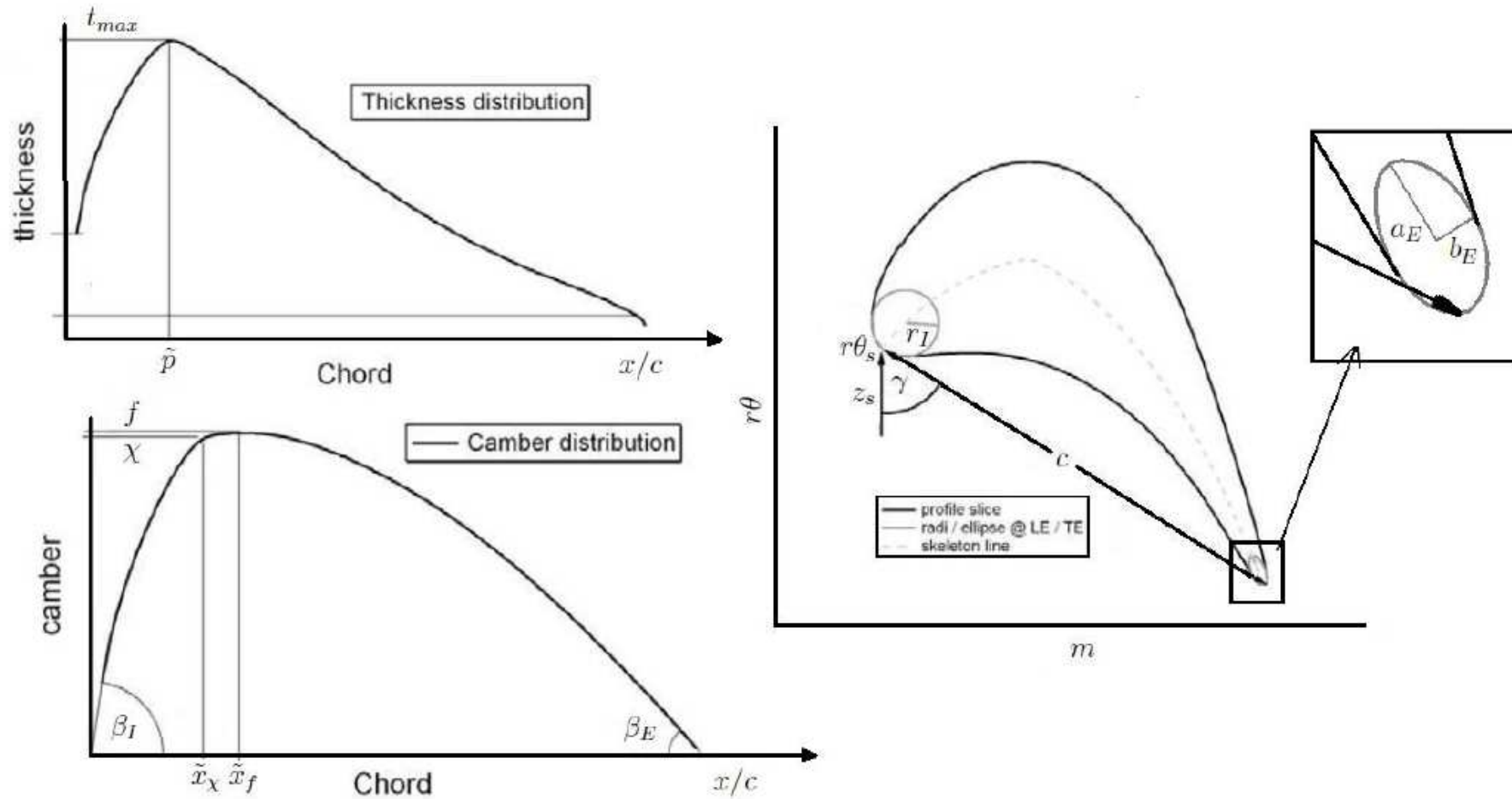
- Back surface diffusion  $d$  is the diffusion from  $M_{max}$  to  $TE$ .
- High shape factors  $H$  indicate flow separation.  
 $H^{TE,SS} < 2$  to insure there is no separation when flow reaches next vane.



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# Aerodynamic blade-to-blade robust turbine blade optimization

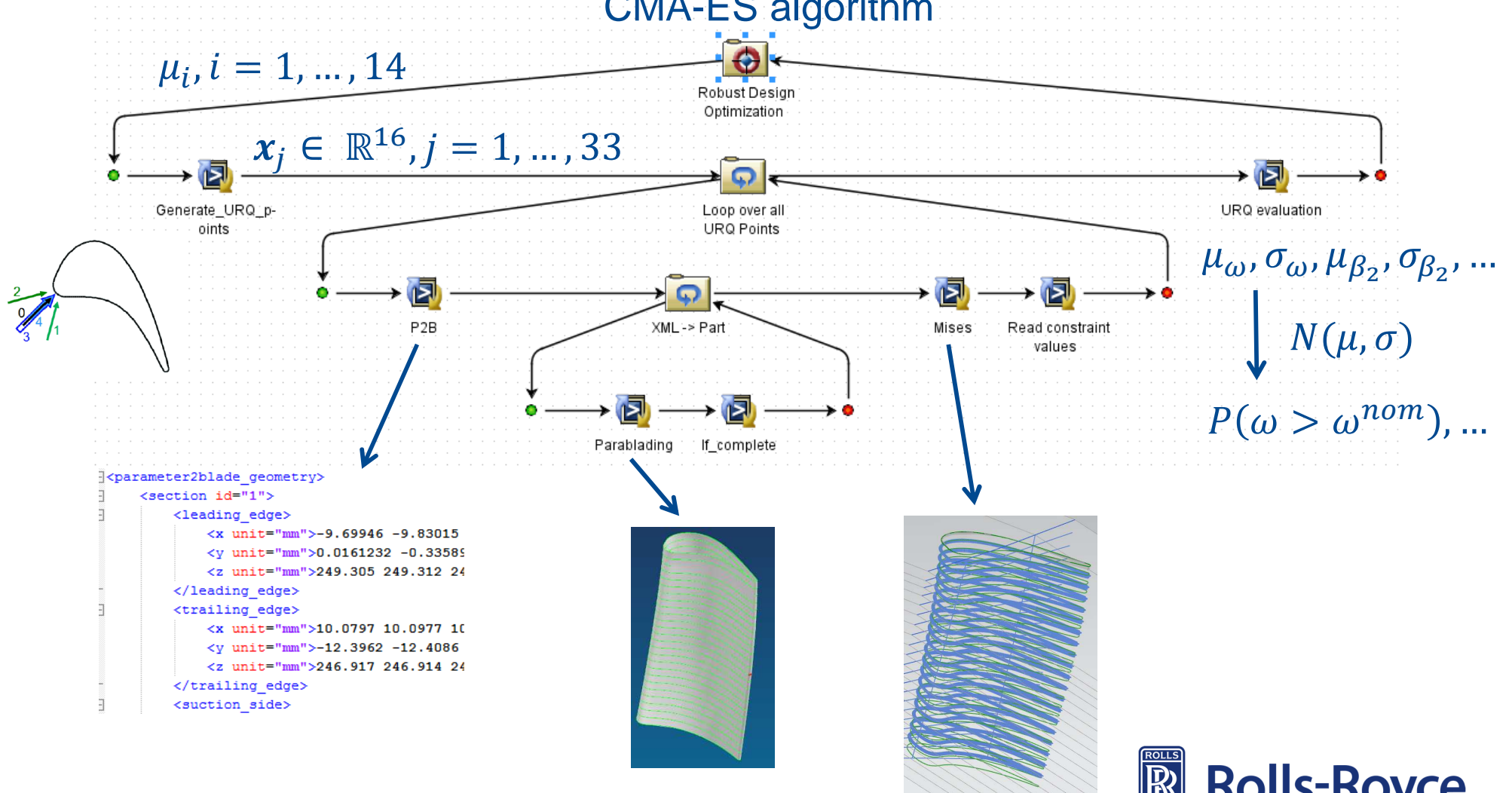
- „NACA“ parameterization:



# Aerodynamic blade-to-blade robust turbine blade optimization

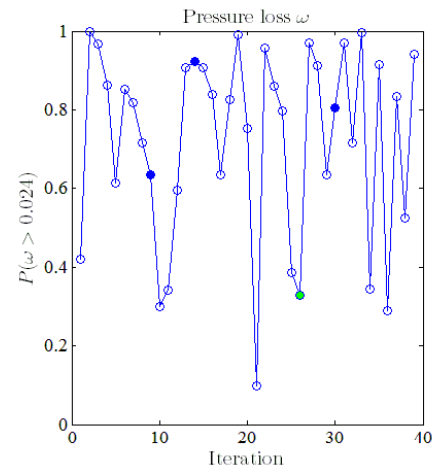
➤ Isight process:

RR customized  
CMA-ES algorithm

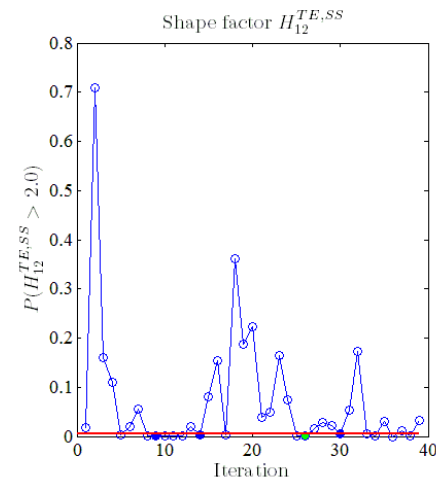
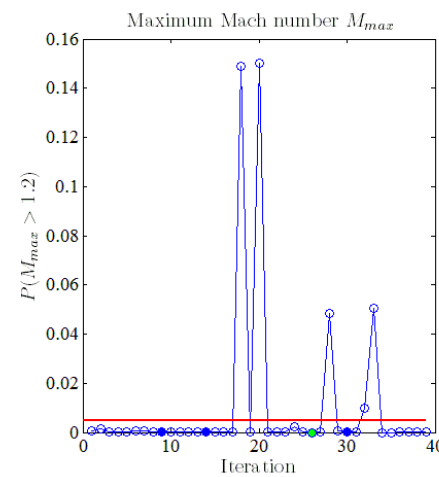
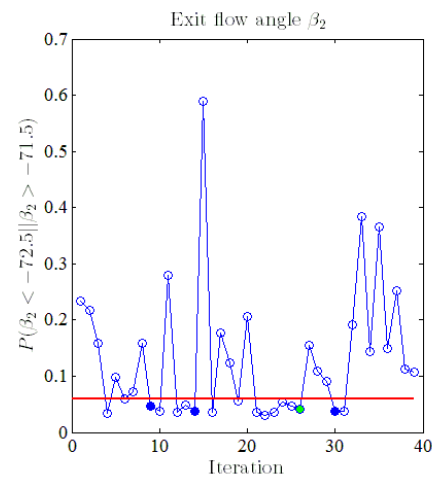
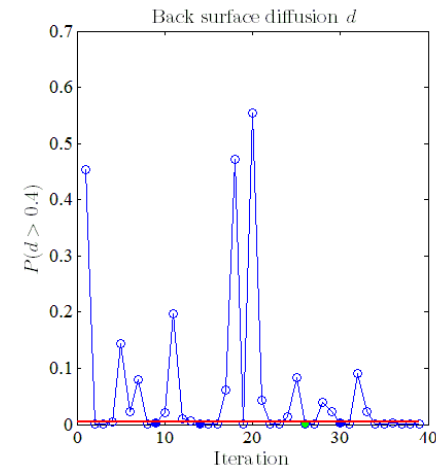


# First results of turbine optimization

- First 39 CMA-ES optimization runs
- Filled dots: All constraints satisfied; Green dot: Optimal solution
- Wall clock time per run: approx. 30 min.



Constraints:



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# Summary of URQ method

- no tuning parameters
- only depending of the first four statistical moments of design parameters
- skewness and kurtosis of design parameters are covered
- correlation between design parameter can be included
- error of order  $O(\sigma_x^4)$
- computationally much cheaper than oLHC



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**Thank you for your attention!**

