

Probabilistic Analysis of Regeneration-Induced Geometry Variances in a Low-Pressure Turbine

8. Dresdner-Probabilistik-Workshop TU Dresden, 8 – 9 October 2015

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Motivation

- **Geometry Variances**
- **Probabilistic Model**

Results

Conclusions

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Outline

Collaborative Research Center (CRC) 871

> Motivation and Objective

> Analysis of Regeneration-Induced Geometry Variances

> Probabilistic Model

Results

Conclusions and Outlook



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CRC 871 - Regeneration of Complex Capital Goods

Scientific basis for maintaining complex capital goods to...

- recondition and improve the functional properties
- refurbish high-value components
- > reduce scrap rates



source: MTU



source: ENERCON



source: Siemens



source: Deutsche Bahn



Project Areas and Subprojects

Probabilistic Analysis of Regeneration-Induced Geometry Variances in a LPT

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Funded by the DFG (German Research Foundation) since 2010

> 2nd funding period (2014 till 2017)



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Motivation and Objective of the Present Study

- > High aerodynamic, mechanical, and thermal loads cause substantial wear
- Regular overhaul and repair of turbine blades
 - → Higher variance after regeneration compared to new engines
 - → Modified aerodynamic and aeroelastic performance
- Efficiency changes in a low pressure turbine (LPT)
 - LPT: $\Delta \eta_{\text{LPT}} = 1\%$ \rightarrow Overall: $\Delta \eta_{\text{overall}} \approx 0.7\%$



Analysis of regeneration induced geometry variances on the aerodynamic performance of a LPT



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Test-Case: Final Stage of a LPT at Cruise



Low Re: Profile aerodynamics dominated by boundary layer transition
 Laminar separation bubble at aft-part of suction side

> High sensitivity to geometrical variances can be expected.



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Determination of Regeneration-Induced Variances

Optical 3D measurements of regenerated turbine blades

Alignment of the measured blades with the reference CAD-model

Extraction of blade profiles over the entire span



Determination of characteristic profile parameters

- Axial chord length
- Stagger angle
- Maximum thickness
- Trailing edge radius

...

(based on Aschenbruck et al. 2013)





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 Image: Image of the image

Measured Blade Geometries

- > Database with 20 regenerated blades
- Data include geometry variances caused by manufacturing, operation, and repair
- > 19 extracted profiles at different span locations of each blade





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Parameterization to Characterize the Profile Geometry

> 12 parameters are used to describe the profile geometry.

Camber line and thickness distribution are modeled by polynomials.





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Geometry Parameter Deviation

Nominal design geometry is used as reference (CAD-model).
Deviations are referred to the parameters of the reference geometry.
Delta-parameter / parameter deviation:

$$\Delta P_{\text{realization}} = P_{\text{realization}} - P_{\text{reference}}$$

Excluding leading edge position and angles, all parameter deviations are normalized with respect to their reference value.





Geometry Parameter Variances at Different Span Locations

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Large scatter range e.g. of leading edge radius and trailing edge angle

Maximum deviation of axial chord and stagger angle at 85% span due to blend repair



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Correlations of Profile Parameter Deviations

Spearman correlation coefficient max * t,max relative) deviation of ΤE 0.5 $\beta_{\rm TE}$ ax 0 с_{max} `c,max LE -0.5 LΕ LE XLE c,max $\begin{bmatrix} \max \\ ax \\ ax \\ \gamma \\ TE \end{bmatrix}$ max max (relative) deviation of ...

Significant correlation e.g. between...

- > stagger angle γ and axial leading edge position x_{LE} (positive)
- > stagger angle γ and axial chord length I_{ax} (negative)
- > axial chord length I_{ax} and axial leading edge position x_{LE} (negative)
- > max. camber c_{max} and circumferential leading edge position $r\theta_{LE}$ (positive)



Scheme of the Probabilistic Model





Tool Chain

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CFD Setup of the Final Stage

Computational domain

- Quasi3D(Q3D)-grid at half-span
- Radial extent: approx. 0.03 l_{ax}
- Each sidewall at constant radius
- Radial resolution: 4 cells
- > Stator: approx. 78.000 cells
- > Rotor: approx. 89.000 cells

> *y*⁺ < 1





Finite volume code TRACE of the DLR

- 2nd order accuracy
- > RANS turbulence closure: Wilcox' (1988) k-@ turbulence model
- Including modification for stagnation point (acc. to Kato and Launder)
- > Non-local correlation-based multimode transition model by Kozulovic (2007)
- Only steady computations performed



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Comparison of the Original and Parameterized Blade Profile



→ Good agreement between the nominal and the parameterized reference



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el. deviation of isentropic efficiency in %

0.2

0.15

0.1

0.05

0

-0.05 - 0

-0.1

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Scatter of Output Parameters

00800

Deviations of isentropic efficiency, flow coefficient and stage loading coefficient are shown relative to the reference.

-0.05 0 0.05 0.1 0.15 0.2 <u>v</u> rel. deviation of stage loading coefficient in %

B CO CO



Scatter range of isentropic efficiency Δη_{is,max} - Δη_{is,min} ≈ 0.25%
 Linear correlation between stage loading and isentropic efficiency
 No significant correlation between flow coefficient and isentropic efficiency



Correlation between Input and Output Parameters



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Significant correlation between trailing edge angle and output parameters No correlation between max. thickness and output parameters



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Conclusions and Outlook

Conclusions

- > Profiles are well characterized by means of 12 geometric parameters.
- > Significant deviations of geometric parameters are found.
- High negative correlation between stagger angle and axial chord length of measured LPT blades

Significant correlation between leading edge angle and

- isentropic efficiency (negative)
- flow coefficient (positive)
- stage loading coefficient (negative)

Geometry variances of most designs lead to an increase in efficiency

Outlook

- Increase the number of measured LPT blades in our database
- > Further improvement of the CFD model and automation of the tool chain
- Analysis of local flow-related parameters
- Variation of operation points



Thank you for your attention!

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